

Space Operations Engineering Through the eyes of an Electronics Engineer

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European Space Operations Centre

Goal of this presentation



- An example of what you can do as an Electronics Engineer at ESA
- Technical overview of the project I am currently working on
- Draw parallels with things you know
- Illustrate everyday problems we encounter
- Show the different aspects of a space mission
- Some things I learnt along the way



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Overview



- My path to space
- Current project at ESA
 - ✓ What is OPS-SAT?
 - ✓ Space Segment
 - ✓ Ground Segment
- Operating a satellite in orbit
 - ✓ Tools we use
 - ✓ Operations concepts
 - ✓ Examples of everyday problems
- Launch and Mission status
- Outlook

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My path to space

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My path to space

- MSc. Electronics-ICT FIIW UHasselt/KUL 2017-2018
 - OSCAR/BEXUS project
 - > Thesis at KUL about CubeSat development
- Instrumentation Engineer at KUL Institute of Physics and Astronomy (9 mo.)
 - > Development of telescope imaging/control systems
 - CubeSat design and development
 - Electronics design
- Spacecraft Operations Engineer at ESA, YGT* (~1,5yrs.)
 - > High-calibre training programme
 - > Pre-launch testing, automation and operations of a space mission





*https://www.esa.int/About_Us/Careers_at_ESA/Graduates_Young_Graduate_Trainees

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ESA in a nutshell



- European Space Agency
- Intergovernmental agency with 22 member states
- Headquartered in Paris
- €14 bn. budget for 2020
- Funding through member state contributions
- Contribution back into member states through contracts with Industry
- Strong focus on
 - ✓ Navigation (Galileo)
 - Earth observation (Copernicus)
 - Science/Astronomy missions

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Current project at ESA

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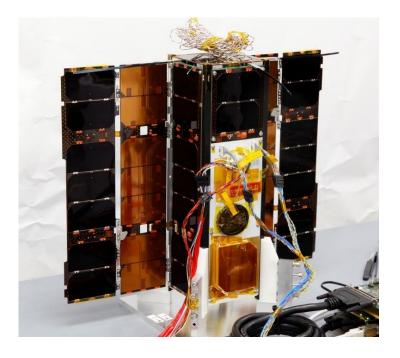
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What is OPS-SAT?

- 3U CubeSat (ESA's first)
- Dedicated to executing hardware/software experiments in Low Earth Orbit
- 100+ companies from 17 countries registered experiments
- Academia, start-ups and large corporations are looking to innovate on OPS-SAT
- Launched 18th Dec 2019 at 05:54:20 from French Guyana (VS23) in a 515km orbit
- Experiments range from telemetry compression algorithms to experimental IP-cores on the FPGA
- Most powerful computer ESA has flown on a satellite





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Space segment (1/2)



- EPS (Electronic Power System)
 - ✓ Solar arrays
 - Array conditioning units
 - ✓ Power regulation
 - ✓ Power storage
- Communications
 - ✓ Radio transceiver
 - Telecommand decoder / Telemetry encoder
 - On-board computer & communications buses
- Attitude control
 - ✓ ADCS (Attitude Determination and Control System)

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Space segment (2/2)

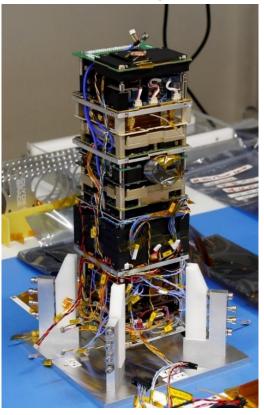
Satellite **bus** (necessary to run the satellite):

- Gomspace UHF AX100 radio + EPS/ACU
- Nanomind A3200 OBC (On-board computer, AVR32)
- S-band (2.2 GHz) TRX TMTC encoder/decoder (256kbps↑ 1Mbps↓)
- GNSS receiver

Satellite **payloads** (experimental systems):

- HD-camera (Nadir-facing)
- Advanced iADCS (Attitude Determination & Control Sys.)
- Optical receiver (data uplink via laser)
- Software Defined Radio (LMS6002D)
- 2x Altera Cyclone V SoC (Dual Core ARM Cortex-A9 + FPGA fabric)
- X-band transmitter (8 GHz, 3-50MBit/s)





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SEPP

- Satellite Experimental Processing Platform, designed by TU Graz
- Cyclone V, Dual Core 800MHz ARM Cortex-A9 (HPS)
- Running embedded Linux 32bit (Ångström)
- Connected via CAN-bus (can0 network interface)
- Nanosat MO Framework (NMF*)
 - ✓ Free open-source ESA SDK on-board to develop applications
 - Abstracts payloads, camera.takePicture(exposure, gain)
- Software is uplinked and installed in the form of IPK files
 - Open package manager (opkg)
 - ✓ Simple to manage; *opkg install*, *opkg remove*, ...
- FPGA portion has IO connections to TMTC encoder/decoder
 - Currently in development for high-speed file transfer

*https://nanosat-mo-framework.github.io





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Software Defined Radio

- Based on LMS6002D
- 300MHz 3.8GHz
- Receive only on OPS-SAT with UHF monopole antenna

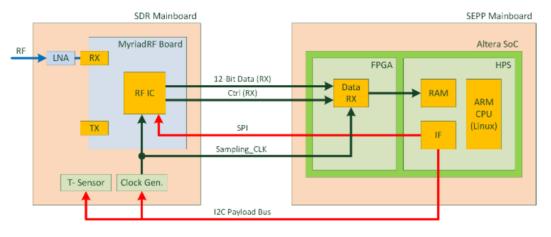
HPS:

Board Monitoring & Configuration via I2C Bus



FPGA:

- RFFE Monitor & Control via SPI Bus
- Data I/O via Parallel IO (FPGA)
- 12 bit I/Q samples

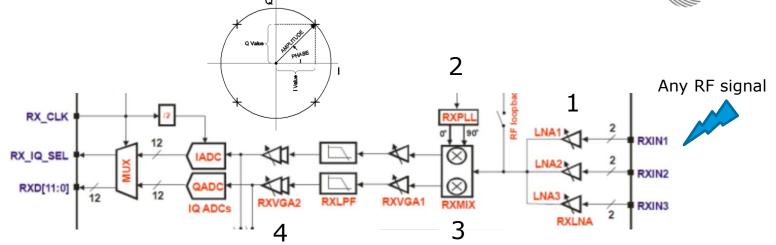


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Software Defined Radio





- 1) Amplify signal with Low Noise Amplifiers
- 2) Set center frequency fc (RXPLL)
- 3) Mix with fc and fc (pi/2) to get I and Q components (downconversion)
- 4) Filter and digitize I and Q components
- 5) Use FPGA IP core to get 12-bit IQ data into RAM/FIFO via HPS bridges

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Software Defined Radio

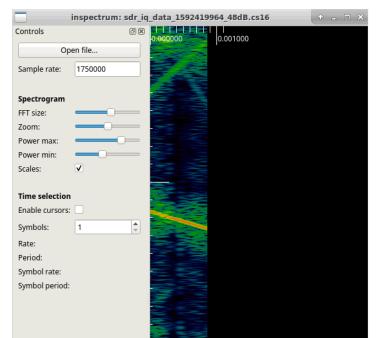


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- 1) Application to write IQ samples to a file (or stream them)
- Analyse frequency components of RF signal FFT(sqrt(I^2+Q^2))

IQ data recorded on-board the satellite while uplinking carrier:



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Ground segment (1/3)



- Communications
 - ✓ Receive telemetry (TM)
 - ✓ Send telecommands (TC)
- Planning
 - ✓ Sequences to execute
 - ✓ Software to run
 - ✓ Manoeuvres to make
- File operations
 - Downlink/uplink of files
 - ✓ Uplink of updates

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Ground segment (2/3)



SMILE LAB at ESOC



ESOC1 3.7m S/X dish

ESOC-2 UHF antenna

- SMILE = Special Mission Infrastructure Laboratory Environment
- All operations automated
- Spacecraft commissioning currently performed remotely due to COVID
- 3 frequency bands:
 - ✓ UHF (437 MHz)
 - S-band (2.3 GHz)
 - X-band (8.1 GHz) \checkmark

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Ground segment (3/3)



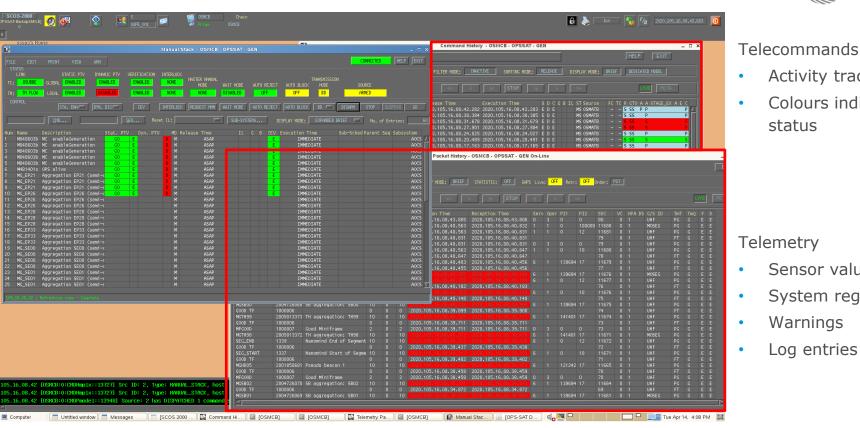
- Core servers run legacy ESA ground software SCOS2000 (used 20+ years)
- Automated using new ESA software (MATIS Mission Automation System)



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Ground segment (3/3)





Activity tracking

Colours indicate

Sensor values

Warnings

Log entries

System registers

status

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Ground segment (3/3)



- Automation of • procedure runtime
- Schedules in XML • form
- Sending TCs ٠
- Checking TM ٠
- Generate reports •
- Send e-mails •
- ECSS-E-ST-70-32C ٠

European Cooperation for Space Standardization
Space Standardization

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• Operating a satellite is similar to being a system administrator

With the differences:

- Your target machine is in space
- Limited window of contact per day (7 minutes per pass, 4 passes per day)
- No debug console(s)
- Hardware failures are fatal
- Limited power available
- Bitflips, latchups and magnetic storms cause software to behave strangely
- Decisions have to be made in minutes

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Do I trust this sensor value?

✓ Understand the route a sensor value goes through in a space mission

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Do I trust this sensor value?



- ✓ Understand the route a sensor value goes through in a space mission
- 1. A sensor value is acquired over the I2C bus (with possible CRC)
- 2. The value gets aggregated in a memory location (Datapool)
- 3. Telemetry process assembles binary payloads from memory locations
- 4. Headers/trailers/CRCs are attached to form CCSDS Space Packets
- 5. Space packets are sent on the main CAN-bus to the TM encoder
- 6. Convolutionally encode the packet into a bitstream
- 7. An RF-transceiver modulates bitstream onto RF-carrier
- 8. The signal travels ~500-1000km through space
- 9. Ground receives and decodes the RF-signal
- 10. A modem decodes the symbols into packets
- 11. Packets are interpreted by the MCS and values extracted

- -> acquisition
- -> caching
- -> fetching
- -> packetization
- -> forwarding
- -> encoding
- -> transmitting
- -> propagation
- -> demodulation
- -> decoding
- -> interpretation

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Reception of occasional close approach alerts by US Space Control

The United States 18th Space Control Squadron has identified a close approach between OPS-SAT (SCC # 44878) and SCC # 23358

Time of Closest Approach: 2020/06/25 13:36:19.000(UTC) Probability of Collision (Pc): 0.000997596 Overall miss distance: 80.0m Radial miss distance (RELATIVE_POSITION_R): 37.2m In-Track miss distance (RELATIVE_POSITION_T): 57.1m Cross-track miss distance (RELATIVE_POSITION_N): -42.6m.

Normal ESA missions:

Check with Flight Dynamics if manoeuvre needs to be made

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Operating a sate

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Reception of occasiona

The United States 18th (SCC # 44878) and SCC

Time of Closest Approa Probability of Collisif Overall miss distance: Radial miss distance (¥ In-Track miss distance Cross-track miss dista

Normal ESA missions:

Check with Flight D





Jonathan O'Callaghan Contributor @

Science

Jonathan is a freelance space journalist that covers commercial spaceflight, space exploration, and astrophysics



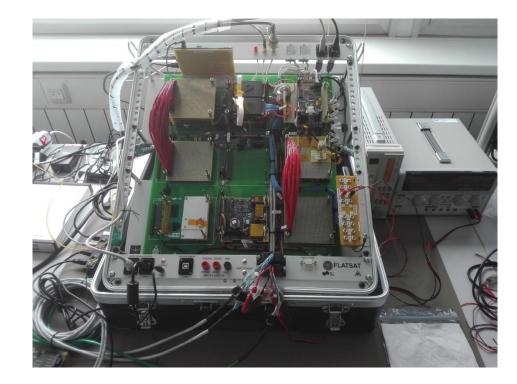


roach between OPS-SAT

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- Procedures, software, sequences and patches are always tested
- Tests happen on ground on the Engineering Model (EM)
- Once validated, a slot is allocated onorbit



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During visibility (in contact with ground):

- Critical checks are performed to quickly check if all systems are nominal
 - Battery voltages and currents
 - ✓ Temperatures
 - ✓ Spin rates
- Hundreds of commands are uplinked with execution times outside of visibility
 - ✓ `Programming' a timeline
 - ✓ 90% of activities happen outside of coverage
- Files are downloaded and software is uploaded

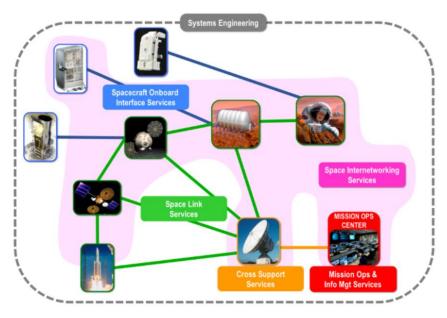
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CCSDS standards

- Definition of telemetry/telecommand formats
- File transfer protocols
- Published as colour coded books





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- Typical 'Blue book'
- Description of packet fields



MASTER CHANNEL ID						
TRANSFER FRAME VERSION NUMBER	SPACECRAFT ID	VIRTUAL CHANNEL ID	OCF FLAG	MASTER CHANNEL FRAME COUNT	VIRTUAL CHANNEL FRAME COUNT	TRANSFER FRAME DATA FIELD STATUS
2 bits	10 bits	3 bits	1 bit			

Figure 4-2: Transfer Frame Primary Header

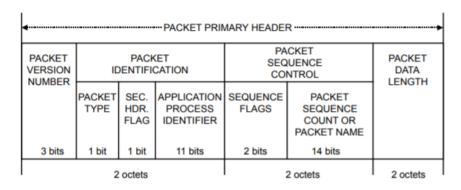


Figure 4-2: Packet Primary Header

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Example problems

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Example problem 1



Statement: Due to an error in requirements, the backup UHF ground station for launch support is not compatible with the RF communication protocol used by the satellite

Solution: Implement and test an SDR modem for the ground station that supports the protocol

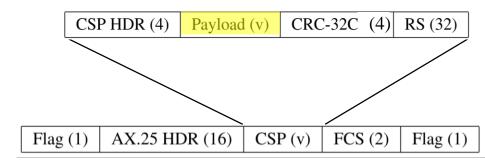
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UHF link



• CubeSat Space Protocol* (CSP) packets + AX.25 + HDLC (nr of bytes)



- Add error correction bytes to the payload
- HDLC uses 'Flags' (0x7E) to indicate packets in a byte sequence
- Bit stuffing ensures this sequence does not occur inside the packet
- Bytes are scrambled using a PRN sequence (scrambler) to ensure enough transitions 0<->1 take place, improves symbol recovery

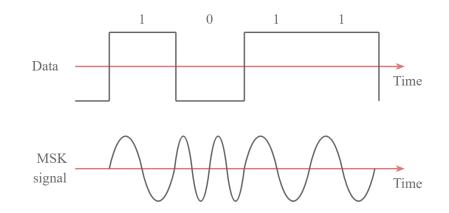
*https://github.com/libcsp/libcsp

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UHF link



- Scrambled bits (now called symbols) are radiated
- GMSK (Gaussian Minimum Shift Keying) @ 9.6kBaud, f=437.2 MHz
 - Bandwidth efficient
 - No phase discontinuities -> less bandwidth than normal FSK



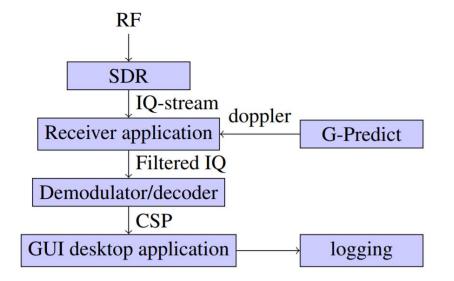
https://www.electronics-notes.com/images/minimum-shift-keying-msk-concept-01.svg

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Implementation in GNU Radio





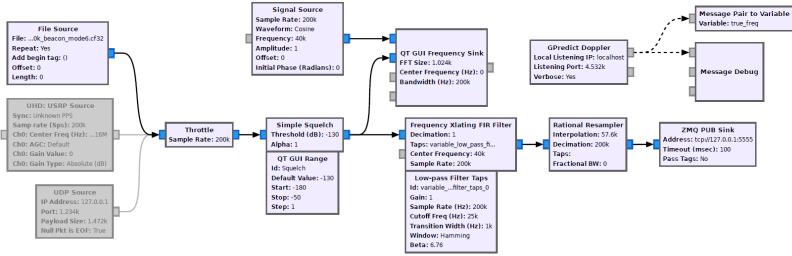
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Receiver



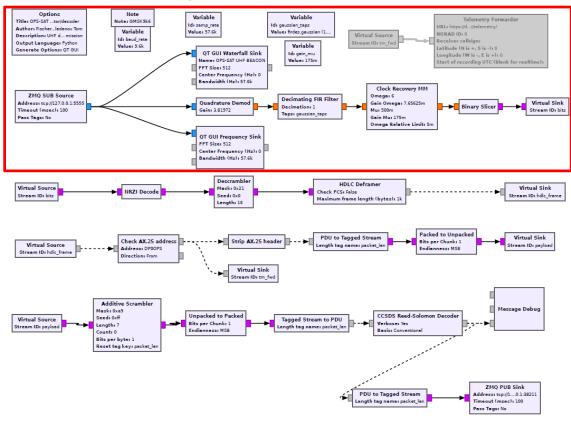


- Simple receiver application
 - Acquire IQ samples (HW f_c set to f_{nom} 40 kHz)
 - Use relative Doppler frequency (40 kHz + f_{Dopp}) for Xlating FIR filter f_c
 - Resample to fixed rate for demodulator/decoder flowgraph
 - Send filtered IQ samples to ZMQ PUB sink

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Demodulator/decoder





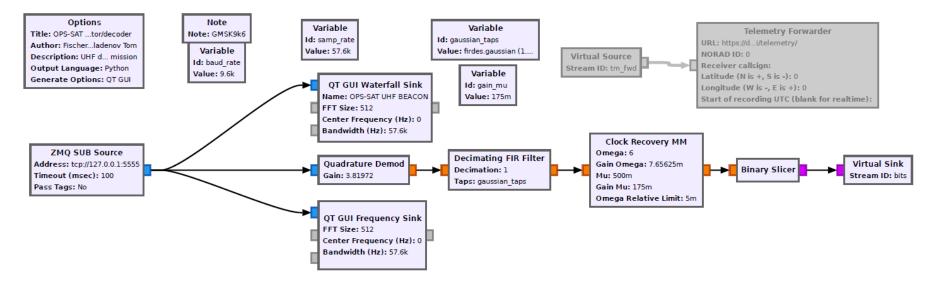
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Demodulator/decoder (1/4)



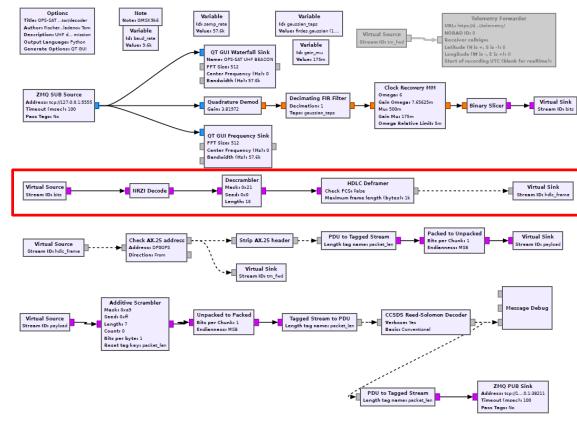


- Quadrature Demod G=2*(samp_rate/baudrate)/pi + Gaussian FIR filter
- Clock Recovery MM + Binary Slicer -> hard symbols

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Demodulator/decoder





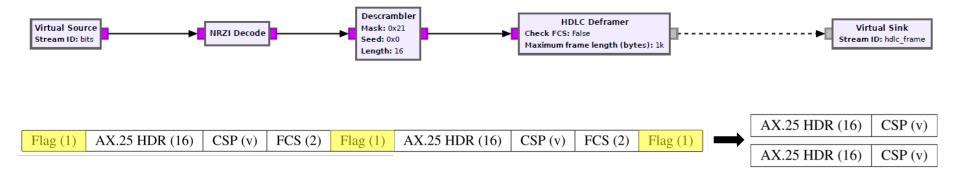
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Demodulator/decoder (2/4)





- NRZI decode + G3RUH descrambling of bitstream
- Deframe 'flagged' HDLC payloads into packetized AX.25

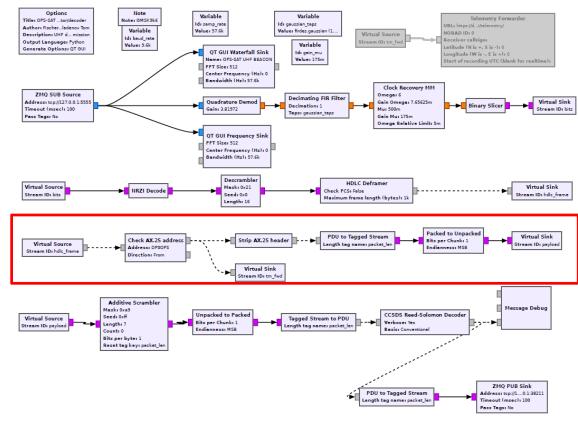
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Demodulator/decoder





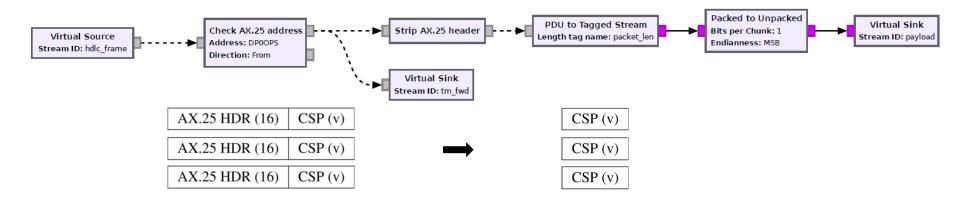
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Demodulator/decoder (3/4)





- Strip AX.25 header and convert to bitstream
- Prepare for decoding of CSP packet

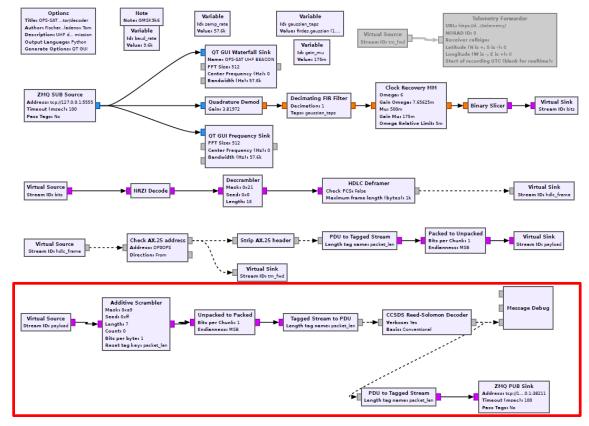
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Demodulator/decoder





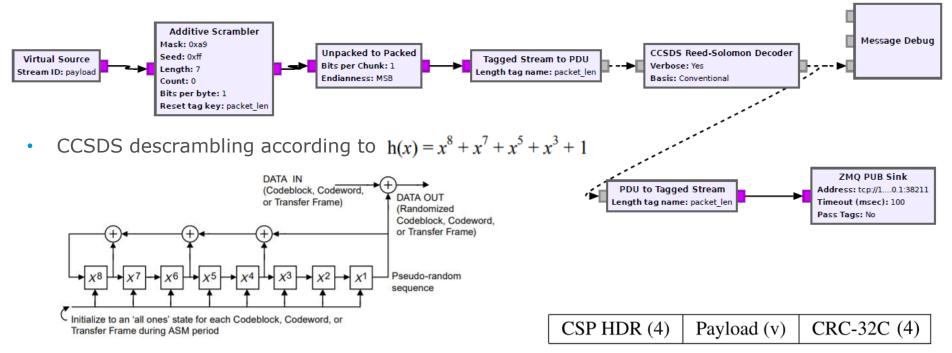
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Demodulator/decoder (4/4)





Reed-Solomon decoding of packet

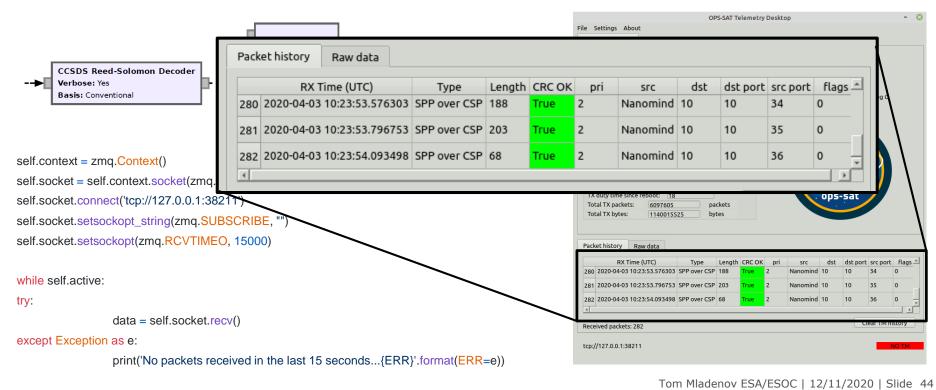
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GUI Application



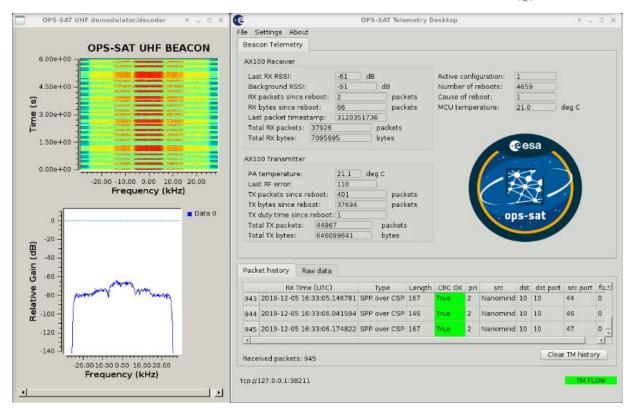
• Viewing the packets from GR in an interactive way



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GUI Application

- GR integration with Python3 and PyQt5
- Use of ZMQ network sockets



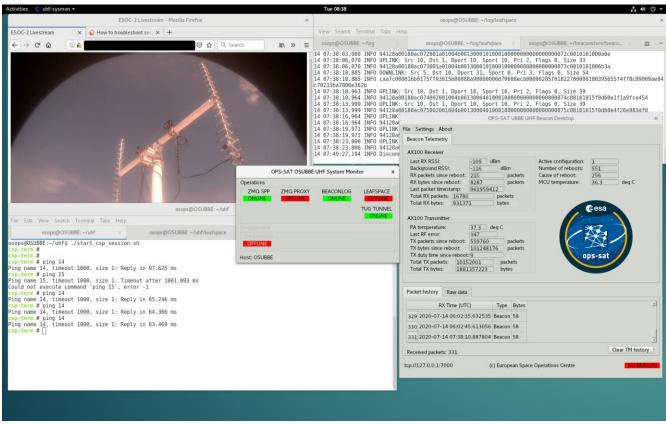
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UHF link



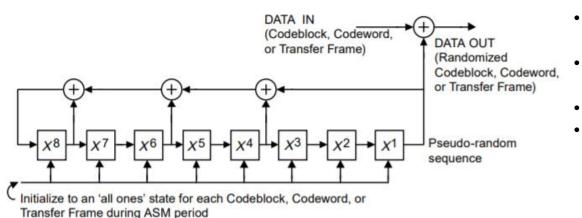


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S-band and X-band

- Use of COTS equipment for decoding on-ground
- COTS limit for X-band = 10MBits/s -> working on SDR solution
- OPS-SAT (and many other satellites) uses the standard CCSDS Space Link
 - Scrambling of data bits using PRN sequence



Scrambling ensures transitions in long sets of 0's and 1's

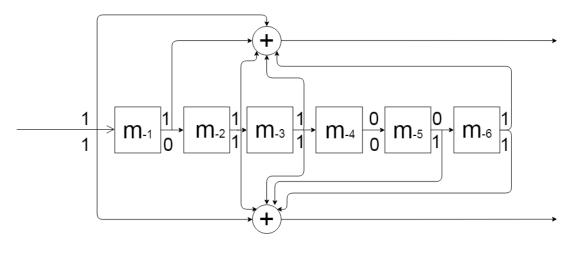
- Easier for ground to perform symbol clock recovery
- Improves spectral efficiency
- Scrambler begin state (seed) is reset at the start of a packet

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S-band and X-band

- Use of COTS equipment for decoding on-ground
- COTS limit for X-band = 10MBits/s -> working on SDR solution
- OPS-SAT (and many other satellites) uses the standard CCSDS Space Link
 - Scrambling of data bits using PRN sequence
 - ✓ Convolutional encoding at R=1/2, K=7 (Voyager code)





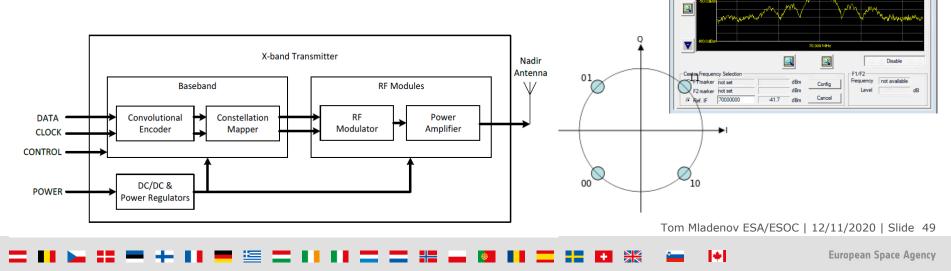


- Apply a sliding polynomial function to bitstream
- 1 error control bit is added to each data bit
- With the same throughput, the overall data rate is halved (R=1/2)
- Significantly improves chances of recovery and error correction

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S-band and X-band

- Use of COTS equipment for decoding on-ground
- COTS limit for X-band = 10MBits/s -> working on SDR solution
- OPS-SAT (and many other satellites) uses the standard CCSDS Space Link
 - Scrambling of data bits using PRN sequence
 - ✓ Convolutional encoding at R=1/2, K=7 (Voyager code)
 - Constellation mapping and RF modulation





Cops-sat.MCS: Video post detection - 1

PLL Unlocked

Channel A Spectrum A Channel B Spectrum B Global



Launch day and Mission status

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- Launch 18th Dec 2019 at 05:54:20 Kourou Local time
- Separation at 14:05 UTC with ANGELS and EyeSAT
- First beacon acquisition by radio amateurs and via SatNOGS
- S/C slow tumble seen in SatNOGS RF-waterfalls
- S/C healthy and plenty of power reserve
- SatNOGS* = Satellite Network Operated Groundstations

*https://db.satnogs.org/

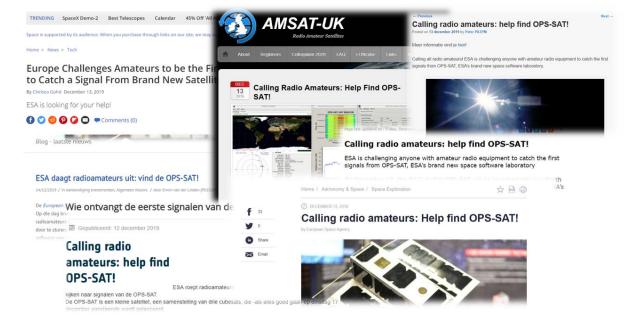


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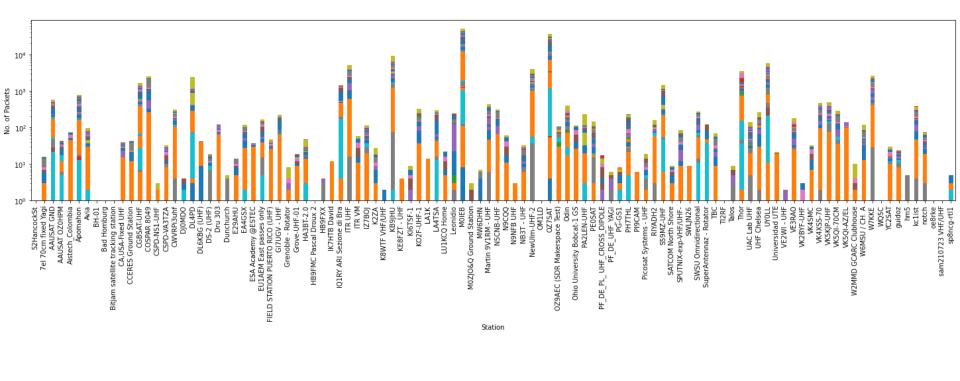
Amateur radio outreach



- Worldwide call for Radio Amateurs to listen for OPS-SAT
- Beacons received and decoded confirming S/C is healthy and functioning before ESA even made contact with the satellite



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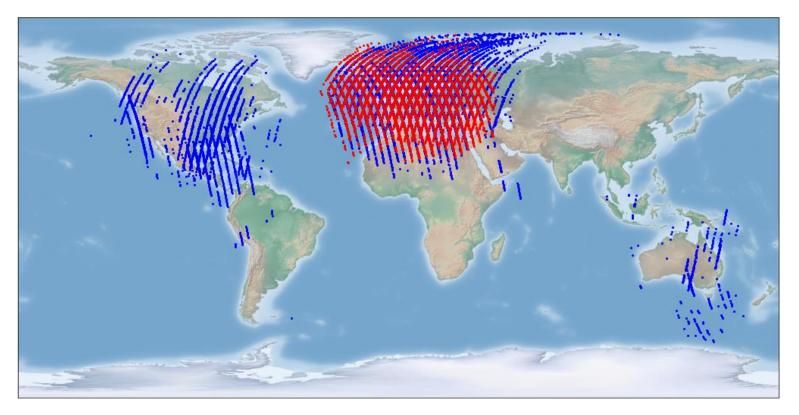


European Space Agency



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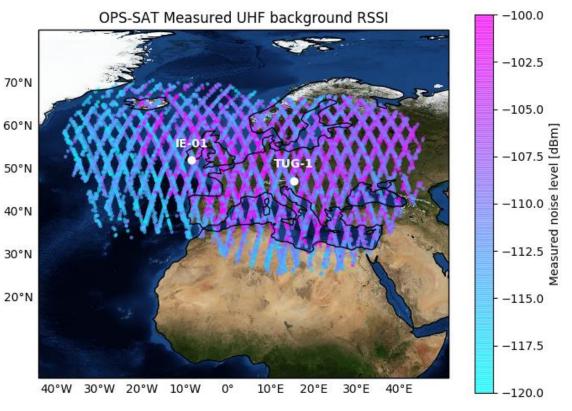
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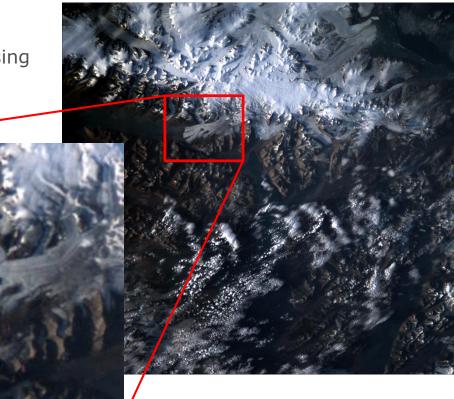
- Initial UHF communications
 were difficult
- Link quite marginal on UHF
- Noise levels 20dB higher over Europe
- Lower Noise levels over the Atlantic and North Africa



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- Commissioned:
 - SEPP (Satellite Experimental Processing Platform)
 - SDR
 - HD-Camera



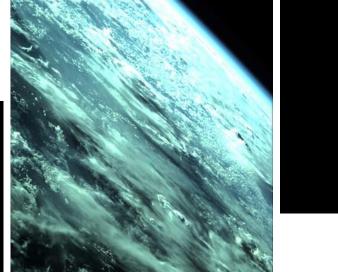


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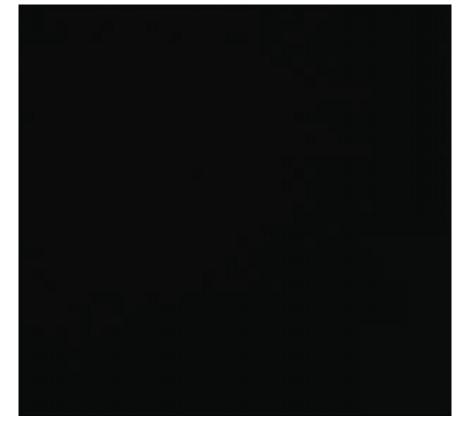


https://www.flickr.com/photos/esa_events/albums/72157716491073681

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Outlook



- Beginning of experimental phase
 - Validation of 'experiments' (validated software packages) on satellite Engineering Model
 - ✓ Uplink to spacecraft
 - ✓ Execution
 - Dissemination of results to experimenter
- External experiments: Start-ups to large space and defence corporations
- Internal experiments by ESA
- Amateur radio experiments over the UHF link
 - Image dissemination
 - RF Uplink experiments

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Thank you for your attention!





Contact: <u>Tom.Mladenov@esa.int</u> Links: <u>https://github.com/esa/gr-opssat</u> <u>https://www.esa.int/Enabling_Support/Operations/OPS-SAT</u> <u>https://public.ccsds.org/default.aspx</u> Personal blog: <u>https://tommladenov.github.io/</u>